



Division of Strength of Materials and Structures

Faculty of Power and Aeronautical Engineering

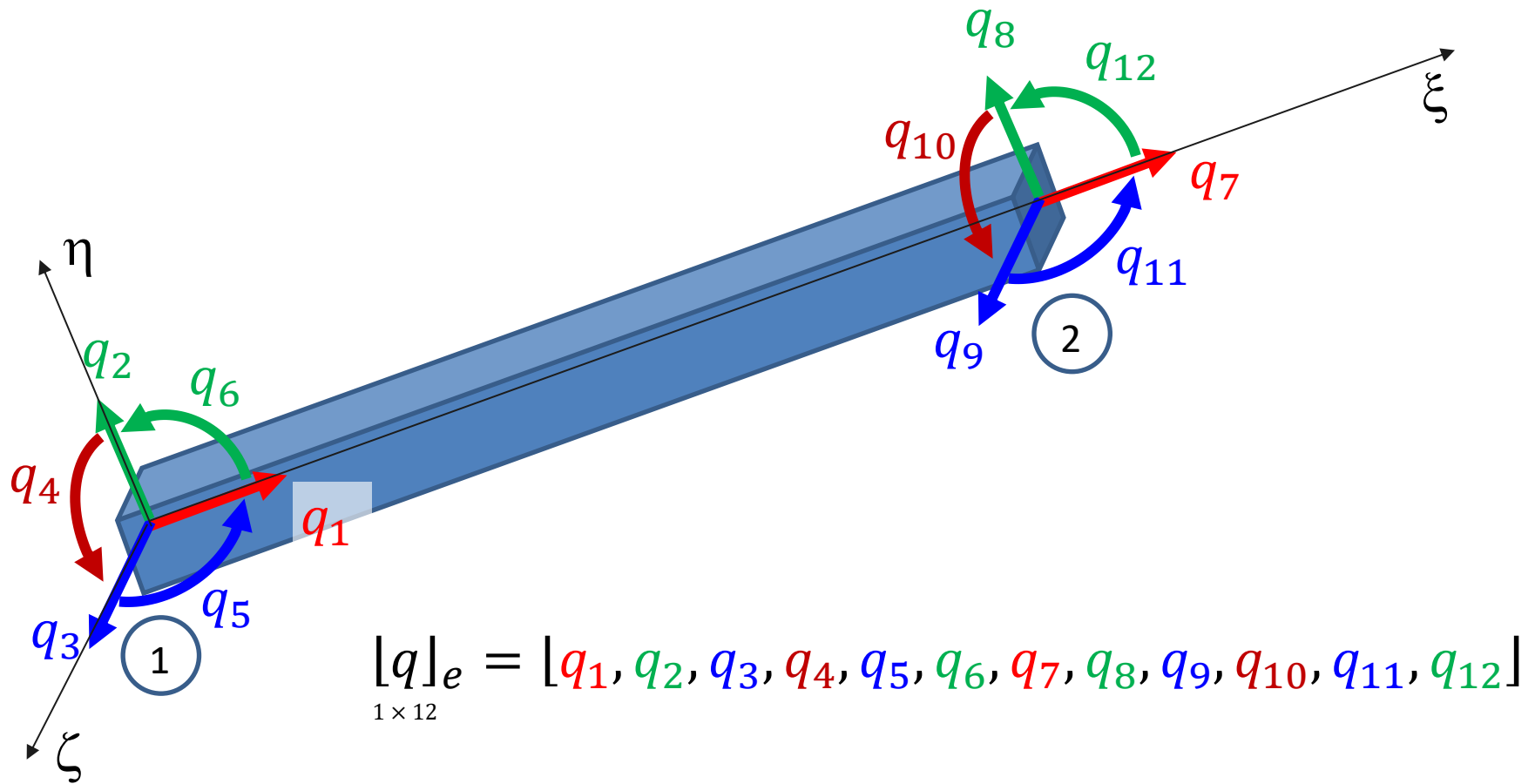


Finite element method (FEM1)

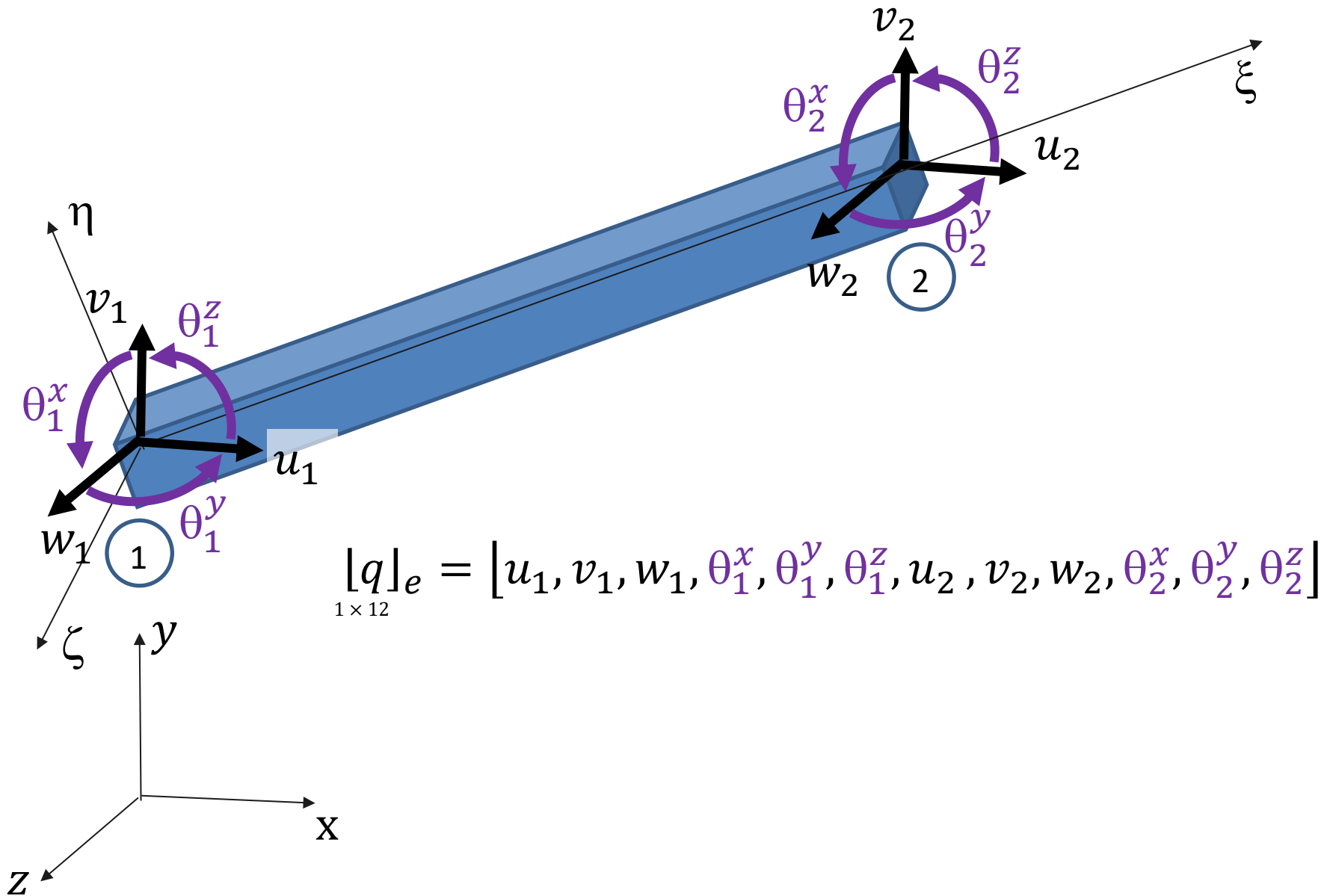
Lecture 11A. 3D Frame element

05.2025

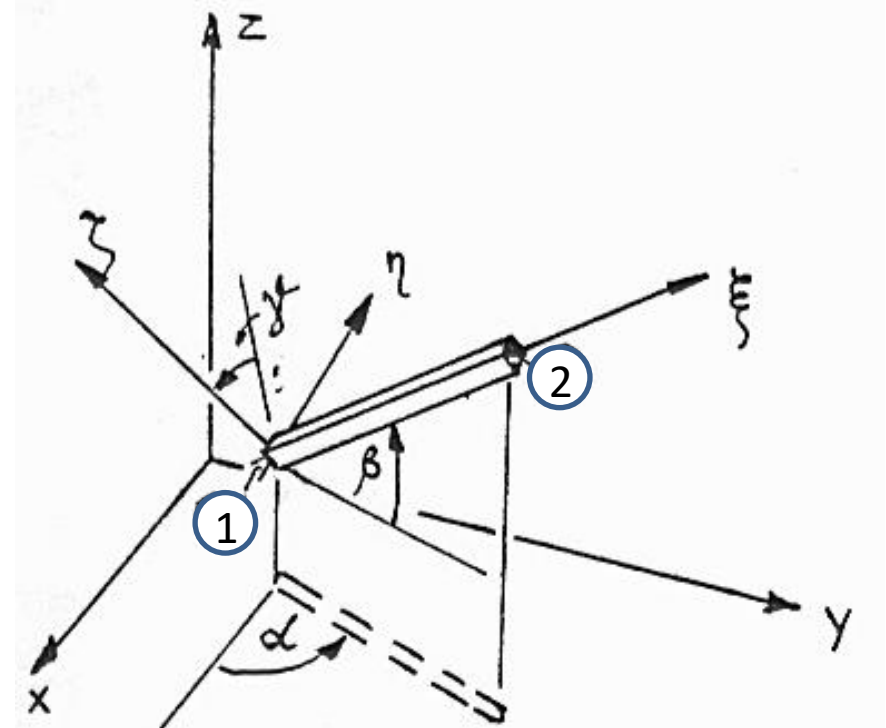
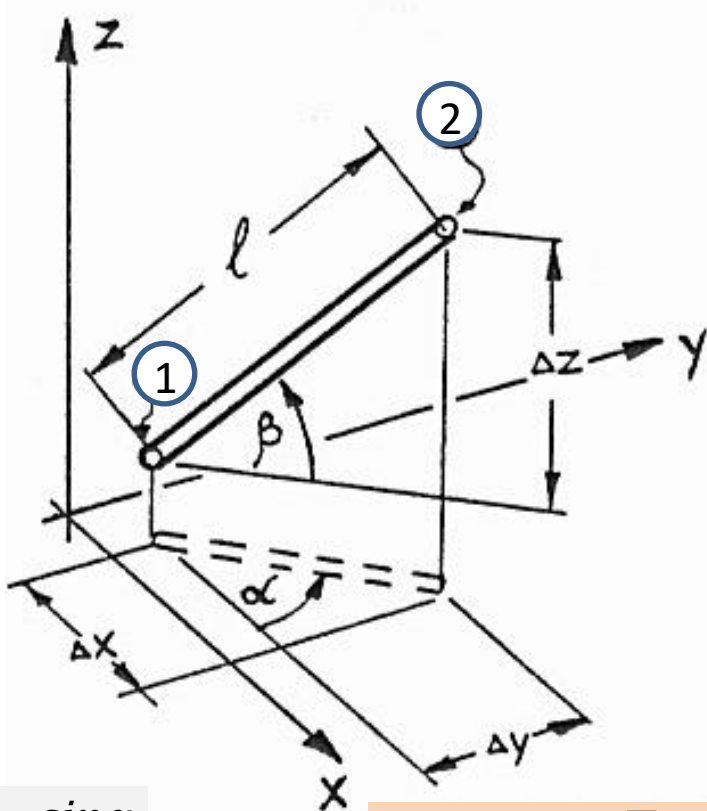
3D Frame element in the local coordinate system (ξ, η, ζ)



3D Frame element in the global coordinate system (x, y, z)



Transformation from the global coordinate system (x, y, z) to the local coordinate system (ξ, η, ζ)

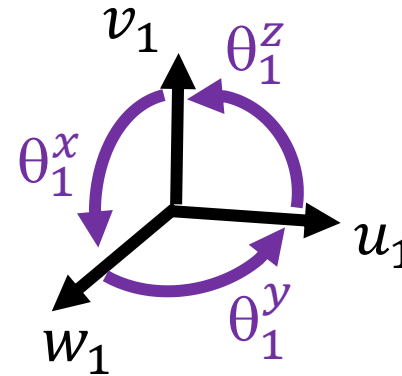
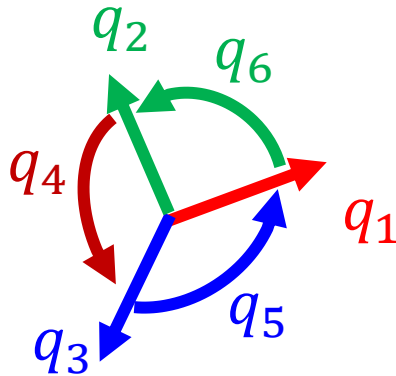


$$\begin{aligned} s\alpha &= \sin\alpha \\ c\alpha &= \cos\alpha \\ s\beta &= \sin\beta \\ c\beta &= \cos\beta \\ s\gamma &= \sin\gamma \\ c\gamma &= \cos\gamma \end{aligned}$$

$$[\xi, \eta, \zeta]^T = [T][x, y, z]^T$$

$$[T] = \begin{bmatrix} c\alpha \cdot c\beta & s\alpha \cdot c\beta & -s\beta \\ -(c\alpha \cdot s\beta \cdot s\gamma + s\alpha \cdot c\gamma) & -(s\alpha \cdot s\beta \cdot s\gamma - c\alpha \cdot c\gamma) & c\beta \cdot s\gamma \\ -(c\alpha \cdot s\beta \cdot c\gamma - s\alpha \cdot s\gamma) & -(s\alpha \cdot s\beta \cdot c\gamma + c\alpha \cdot s\gamma) & c\beta \cdot c\gamma \end{bmatrix}$$

Transformation of degrees of freedom from the global coordinate system (x, y, z)
to local coordinate system (ξ, η, ζ)



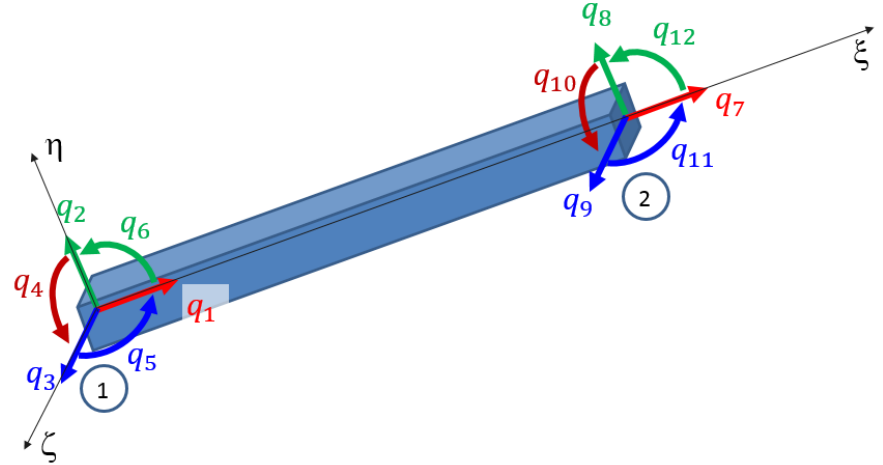
$$\{q\}_e = \begin{bmatrix} [T] \\ [T] \end{bmatrix} \{q_g\}_e = [T_r] \{q_g\}_e$$

$$\begin{aligned} s\alpha &= \sin\alpha \\ c\alpha &= \cos\alpha \\ s\beta &= \sin\beta \\ c\beta &= \cos\beta \\ s\gamma &= \sin\gamma \\ c\gamma &= \cos\gamma \end{aligned}$$

$$[T] = \begin{bmatrix} c\alpha \cdot c\beta & s\alpha \cdot c\beta & -s\beta \\ -(c\alpha \cdot s\beta \cdot s\gamma + s\alpha \cdot c\gamma) & -(s\alpha \cdot s\beta \cdot s\gamma - c\alpha \cdot c\gamma) & c\beta \cdot s\gamma \\ -(c\alpha \cdot s\beta \cdot c\gamma - s\alpha \cdot s\gamma) & -(s\alpha \cdot s\beta \cdot c\gamma + c\alpha \cdot s\gamma) & c\beta \cdot c\gamma \end{bmatrix}$$

Frame stiffness matrix components

$$[k_N] = \begin{bmatrix} \frac{AE}{l} & \frac{-AE}{l} \\ \frac{-AE}{l} & \frac{AE}{l} \end{bmatrix}$$



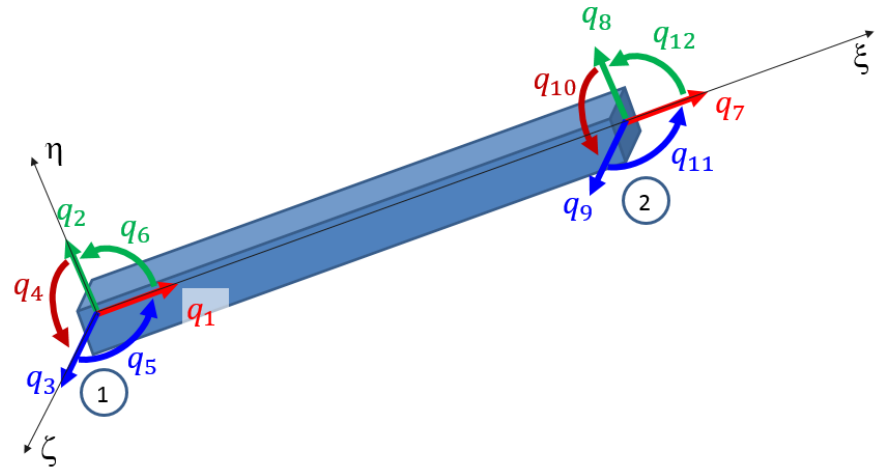
Stretching dependent on $[q_1, q_7]$ degrees of freedom

$$[k_S] = \begin{bmatrix} \frac{GJ_S}{l} & \frac{-GJ_S}{l} \\ \frac{-GJ_S}{l} & \frac{GJ_S}{l} \end{bmatrix}$$

Twisting dependent on $[q_4, q_{10}]$ degrees of freedom

Frame stiffness matrix components

$$\left[k_{M_g \eta} \right] = \begin{bmatrix} \frac{12EJ_\eta}{l^3} & \frac{-6EJ_\eta}{l^2} & \frac{-12EJ_\eta}{l^3} & \frac{-6EJ_\eta}{l^2} \\ \frac{-6EJ_\eta}{l^2} & \frac{4EJ_\eta}{l} & \frac{6EJ_\eta}{l^2} & \frac{2EJ_\eta}{l} \\ \frac{-12EJ_\eta}{l^3} & \frac{6EJ_\eta}{l^2} & \frac{12EJ_\eta}{l^3} & \frac{6EJ_\eta}{l^2} \\ \frac{-6EJ_\eta}{l^2} & \frac{2EJ_\eta}{l} & \frac{6EJ_\eta}{l^2} & \frac{4EJ_\eta}{l} \end{bmatrix}$$

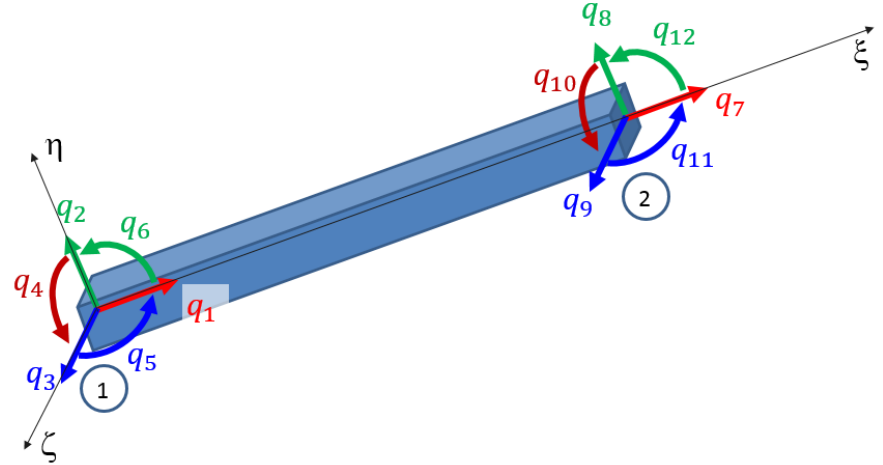


Bending about η axis dependent on $[q_3, q_5, q_9, q_{11}]$ degrees of freedom

$$\left[k_{M_g \zeta} \right] = \begin{bmatrix} \frac{12EJ_\zeta}{l^3} & \frac{6EJ_\zeta}{l^2} & \frac{-12EJ_\zeta}{l^3} & \frac{6EJ_\zeta}{l^2} \\ \frac{6EJ_\zeta}{l^2} & \frac{4EJ_\zeta}{l} & \frac{-6EJ_\zeta}{l^2} & \frac{2EJ_\zeta}{l} \\ \frac{-12EJ_\zeta}{l^3} & \frac{-6EJ_\zeta}{l^2} & \frac{12EJ_\zeta}{l^3} & \frac{-6EJ_\zeta}{l^2} \\ \frac{6EJ_\zeta}{l^2} & \frac{2EJ_\zeta}{l} & \frac{-6EJ_\zeta}{l^2} & \frac{4EJ_\zeta}{l} \end{bmatrix}$$

Bending about ζ axis dependent on $[q_2, q_6, q_8, q_{12}]$ degrees of freedom

Elastic strain energy of the frame element



Elastic strain energy of an element:

$$U_e = \frac{1}{2} [q]_e [k]_e \{q\}_e = \frac{1}{2} [q_g]_e [T_r]^T [k]_e [T_r] \{q_g\}_e,$$

$$U_e = \frac{1}{2} [q_g]_e [k^g]_e \{q_g\}_e,$$

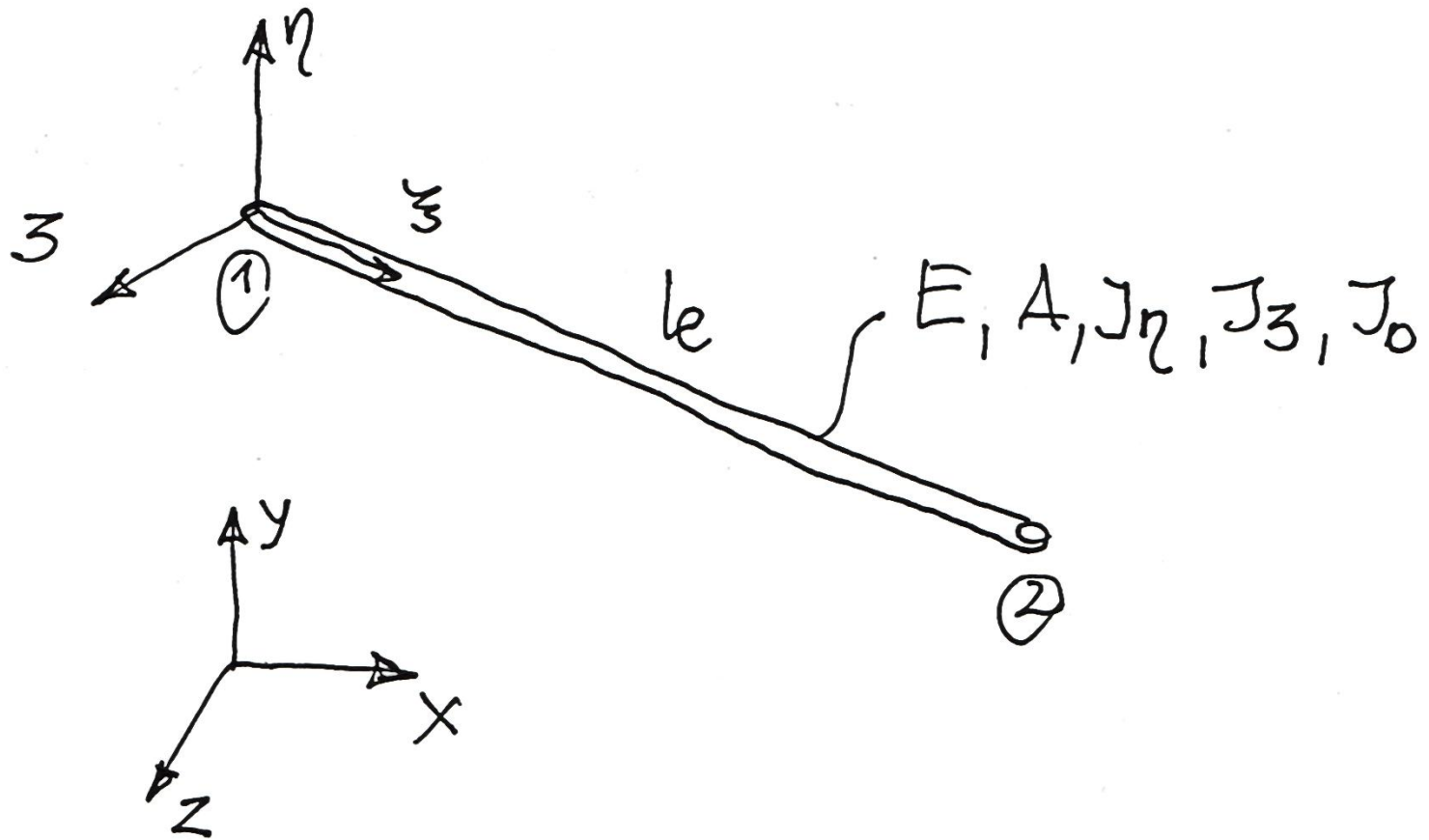
Element stiffness matrix:

$$[k^g]_e = [T_r]^T [k]_e [T_r]$$

3D Frame element stiffness matrix in local coordinate system (ξ, η, ζ)

[illegible]

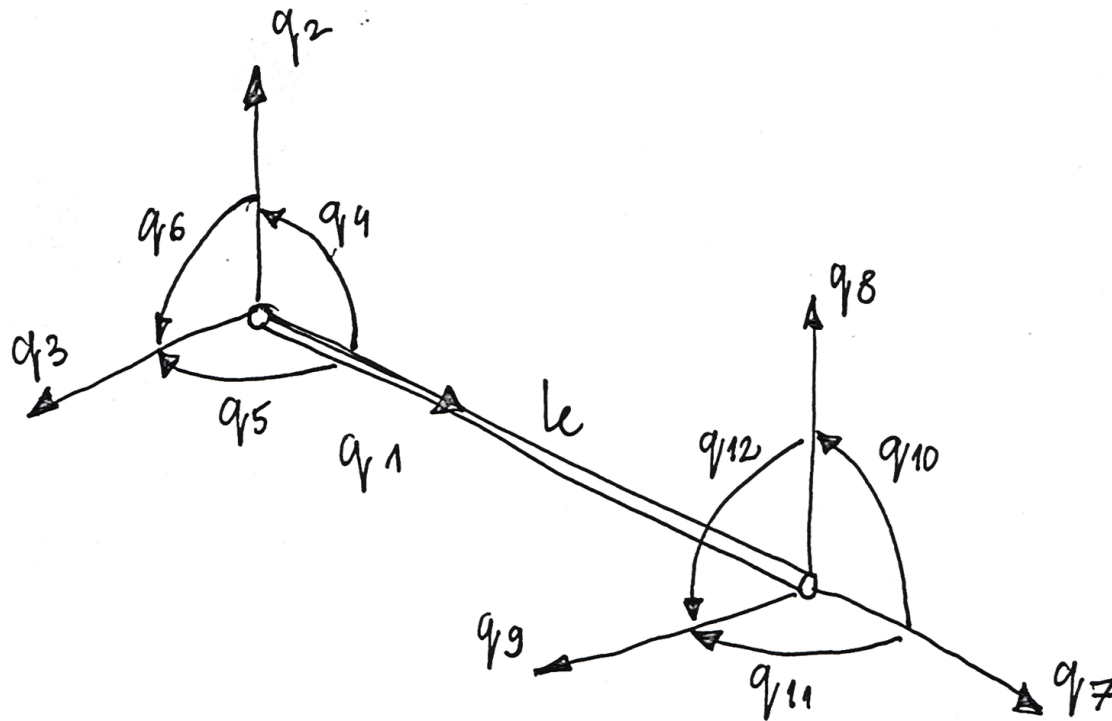
3D FRAME ELEMENT



LOCAL PARAMETERS IN THE COORDINATE SYSTEM ξ_2

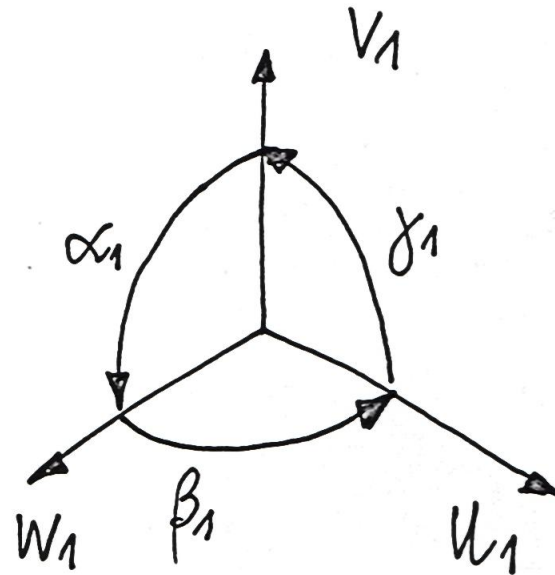
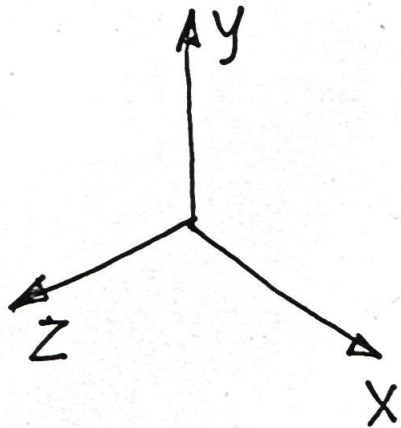
$${}^Lq_e = [q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}]$$

1×12



Local parameters in the coordinate system xyz

$${}_{1 \times 12} [q_g]_e = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2]$$



$$U_e = \underset{1 \times 12}{\frac{1}{2} L} \underset{12 \times 12}{\cancel{q}} \underset{12 \times 1}{[k]}_e \{ \underset{12 \times 1}{q} \}_e \quad \text{where :}$$

$$[k]_e =$$

a						-a					
	b		d				-b		d		
		c		e				-c		e	
	d		2r				-d		r		
		e		2s				-e		s	
					t						-t
-a						a					
	-b		-d				b		-d		
		-c		-e				c		-e	
	d		r				-d		2r		
		e		s				-e		2s	
					-t						t

$$a = \frac{EA}{L}, \quad b = \frac{12EJ_3}{l_e^3}, \quad c = \frac{12EJ_2}{l_e^3}, \quad d = \frac{6EJ_3}{l_e^2},$$

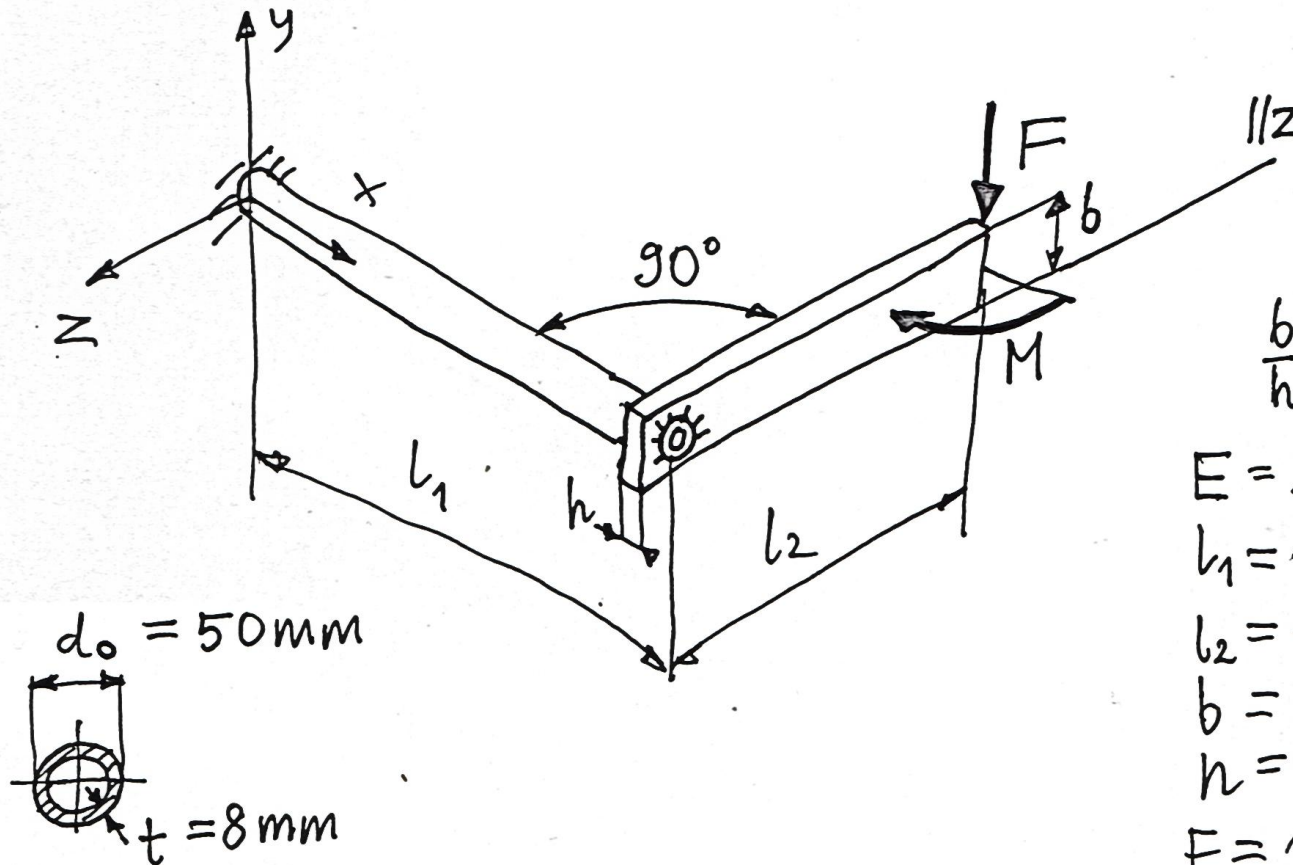
$$e = \frac{6EJ_2}{l_e^2}, \quad r = \frac{2EJ_3}{l_e}, \quad s = \frac{2EJ_2}{l_e}, \quad t = \frac{G \cdot J_0}{l_e}$$

$$U_e = \underbrace{\frac{1}{2} L q_g^T e \cdot [T_f]^T [k]_e [T_f] \cdot \{q_g\}_e}_{[k_g]_e}$$

$1 \times 12 \quad 12 \times 12 \quad 12 \times 12 \quad 12 \times 12 \quad 12 \times 1$

$[k_g]_e$
 12×12

EXAMPLE : BUILD A FE MODEL USING 3D FRAME ELEMENTS. FIND UNKNOWN DISPLACEMENTS, STRESSES AND REACTIONS.



$$\frac{b}{h} = 2$$

$$E = 2 \cdot 10^5 \text{ MPa}$$

$$l_1 = 1200 \text{ mm}$$

$$l_2 = 750 \text{ mm}$$

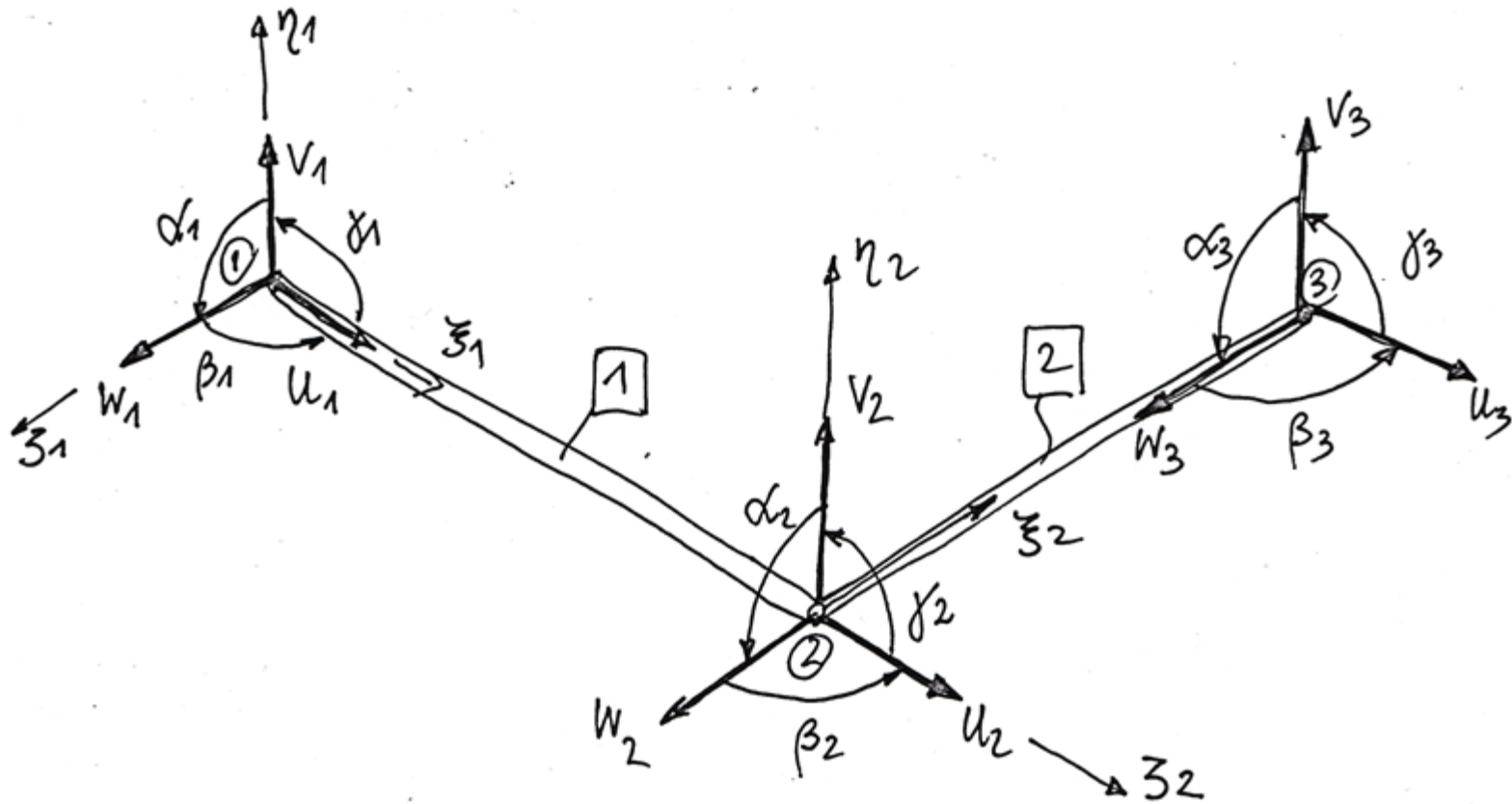
$$b = 60 \text{ mm}$$

$$h = 30 \text{ mm}$$

$$F = 1000 \text{ N}$$

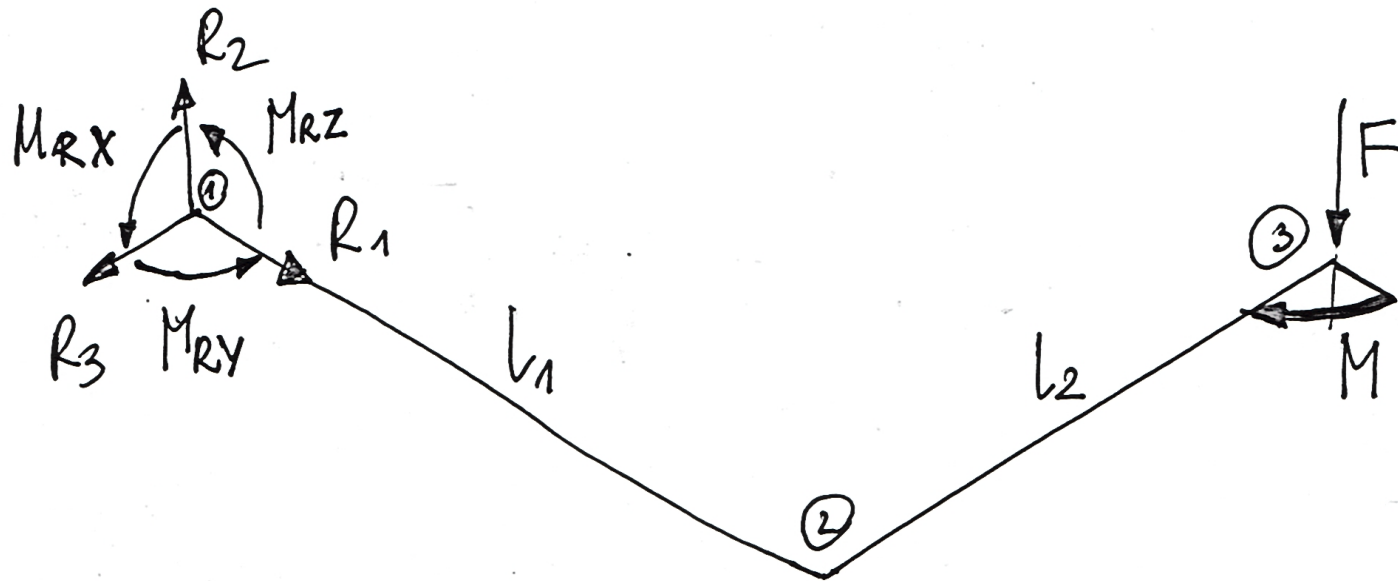
$$M = 1000000 \text{ Nmm}$$

Global parameters in the coordinate system xyz



$$\begin{matrix} \mathbf{[q]} \\ 1 \times 18 \end{matrix} = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2, u_3, v_3, w_3, \alpha_3, \beta_3, \gamma_3]$$

Global load vector



$$[F] = [R_1, R_2, R_3, M_{Rx}, M_{Ry}, M_{Rz}, 0, 0, 0, 0, 0, 0, -F, 0, 0, -M, 0]$$

1×18

ELEMENT 1:

$$[q]_1 = [q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}]$$

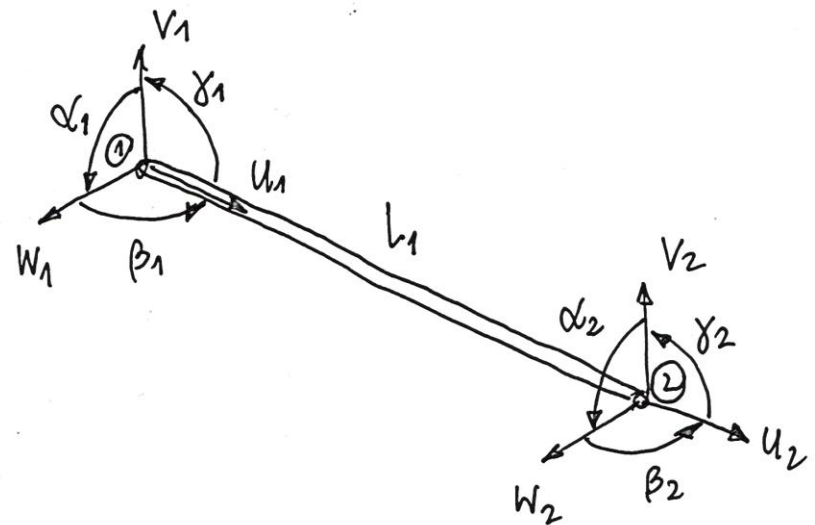
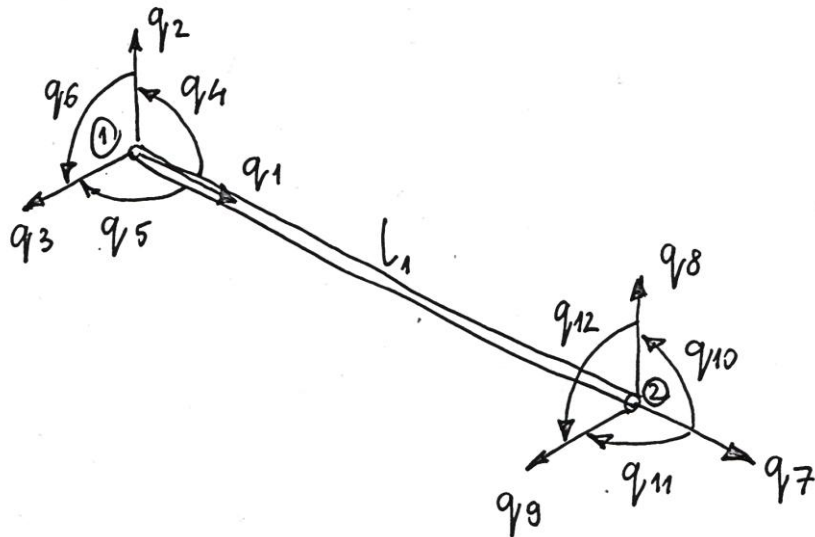
1×12

$$[q]_1 = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2]$$

1×12

$$q_1 = u_1, \quad q_2 = v_1, \quad q_3 = w_1, \quad q_4 = \gamma_1, \quad q_5 = -\beta_1, \quad q_6 = \alpha_1$$

$$q_7 = u_2, \quad q_8 = v_2, \quad q_9 = w_2, \quad q_{10} = \gamma_2, \quad q_{11} = -\beta_2, \quad q_{12} = \alpha_2$$



$$\underbrace{\{q\}_1}_{12 \times 1} = \underbrace{[T_f]_1}_{12 \times 12} \cdot \underbrace{\{q_g\}_1}_{12 \times 1}$$

$$\underbrace{[T_f]_1}_{12 \times 12} = \left[\begin{array}{ccc|ccc} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ \hline & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ \hline & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \end{array} \right] = [T_f]^T$$

$\begin{bmatrix} 0 \end{bmatrix}_{6 \times 6}$
 $\begin{bmatrix} 0 \end{bmatrix}_{6 \times 6}$

$$[k_g]_1 = [T_f]_1^T \cdot [k]_1 \cdot [T_f]_1$$

12×12 12×12 12×12 12×12

$$a_1 = \frac{EA_1}{L_1}, \quad b_1 = \frac{12EJ_{31}}{L_1^3}, \quad c_1 = b_1, \quad d_1 = \frac{6EJ_{31}}{L_1^2}, \quad e_1 = d_1,$$

$$r_1 = \frac{2EJ_{31}}{L_1}, \quad s_1 = r_1, \quad t_1 = \frac{GJ_{01}}{L_1}, \quad A_1 = \frac{\pi(d_o^2 - (d_o - 2t)^2)}{4}$$

$$J_{31} = J_{r1} = \frac{\pi}{64} (d_o^4 - (d_o - 2t)^4), \quad J_{01} = 2 \cdot J_{31}$$

$$[k_g]_1^* = \begin{bmatrix} [k_g]_1 & [0] \\ [0] & [0] \end{bmatrix}$$

18×18 12×6 6×12 6×6

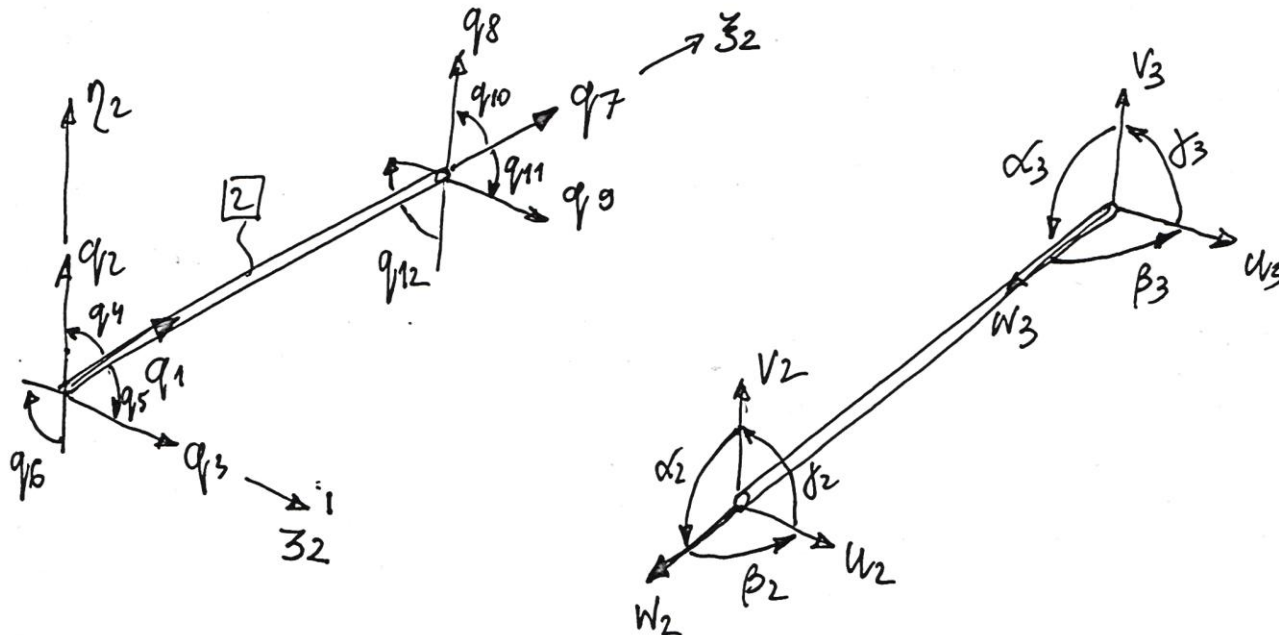
ELEMENT [2]

$$[q]_2 = [q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}]$$

$$[q]_2 = [u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2, u_3, v_3, w_3, \alpha_3, \beta_3, \gamma_3]$$

$$q_1 = -w_2, \quad q_2 = v_2, \quad q_3 = u_2, \quad q_4 = \alpha_2, \quad q_5 = -\beta_2, \quad q_6 = -\gamma_2$$

$$q_7 = -w_3, \quad q_8 = v_3, \quad q_9 = u_3, \quad q_{10} = \alpha_3, \quad q_{11} = -\beta_3, \quad q_{12} = -\gamma_3$$



$$\underbrace{\{q\}_2}_{12 \times 1} = \underbrace{[T_f]_2}_{12 \times 12} \cdot \underbrace{\{q_9\}_2}_{12 \times 1}$$

$$[T_f]_2 = \left[\begin{array}{cccccccccccc} 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 \\ \hline & & & & & & 0 & 0 & -1 & 0 & 0 & 0 \\ & & & & & & 0 & 1 & 0 & 0 & 0 & 0 \\ & & & & & & 1 & 0 & 0 & 0 & 0 & 0 \\ & & & & & & 0 & 0 & 0 & 1 & 0 & 0 \\ & & & & & & 0 & 0 & 0 & 0 & -1 & 0 \\ \hline & & & & & & 0 & 0 & 0 & 0 & 0 & -1 \end{array} \right]$$

$\begin{bmatrix} 0 \end{bmatrix}$
 6×6

$$\begin{matrix} [k_g]_2 & = & [T_f]_2^T & \cdot & [k]_2 & \cdot & [T_f]_2 \\ 12 \times 12 & & 12 \times 12 & & 12 \times 12 & & 12 \times 12 \end{matrix}$$

$$a_2 = \frac{EA_2}{l_2}, \quad b_2 = \frac{12EJ_{32}}{l_2^3}, \quad c_2 = \frac{12EJ_{\eta 2}}{l_2^3}, \quad d_2 = \frac{6EJ_{32}}{l_2^2}$$

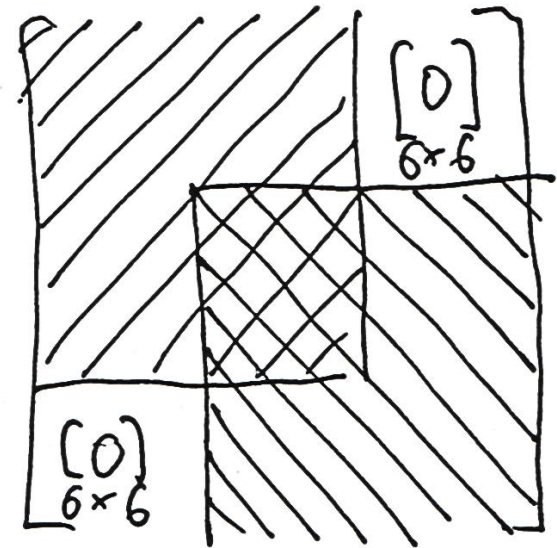
$$e_2 = \frac{6EJ_{\eta 2}}{l_2^2}, \quad f_2 = \frac{2EJ_{32}}{l_2}, \quad g_2 = \frac{2EJ_{\eta 2}}{l_2}, \quad t_2 = \frac{G \cdot J_{02}}{l_2}$$

$$J_{32} = \frac{hb^3}{12}, \quad J_{\eta 2} = \frac{bh^3}{12}, \quad J_{02} = 0.457 bh^3$$

$$A_2 = b \cdot h$$

$$[k_g]_2^* = \begin{bmatrix} \begin{matrix} [0] \\ 6 \times 6 \end{matrix} & \begin{matrix} [0] \\ 6 \times 12 \end{matrix} \\ \begin{matrix} [0] \\ 12 \times 6 \end{matrix} & \begin{matrix} [k_g]_2 \\ \text{shaded} \end{matrix} \end{bmatrix}$$

$$\begin{array}{c}
 [K] \\
 18 \times 18
 \end{array}
 =
 \begin{array}{c}
 [K_g]_1^* \\
 18 \times 18
 \end{array}
 +
 \begin{array}{c}
 [K_g]_2^* \\
 18 \times 18
 \end{array}
 =$$

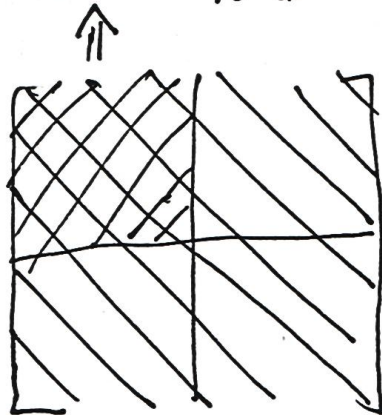


$$\underset{18 \times 18}{[K]} \cdot \underset{18 \times 1}{\{q\}} = \underset{18 \times 1}{\{F\}}$$

$$u_1 = 0, u_4 = 0, w_1 = 0$$

$$/ \quad \alpha_1 = 0, \beta_1 = 0, \gamma_1 = 0$$

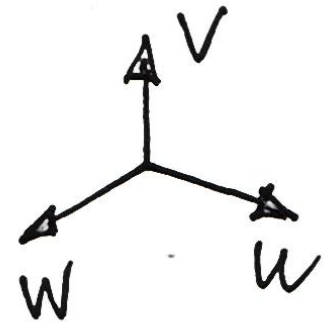
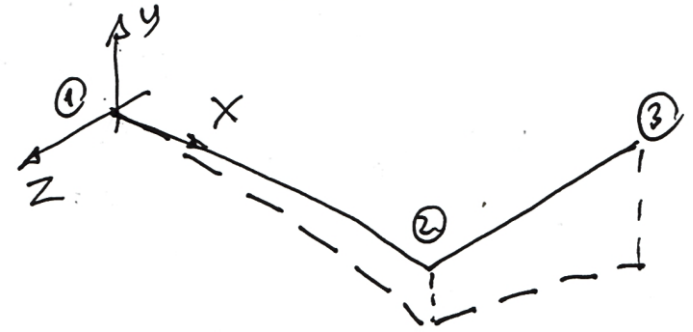
$$\underset{12 \times 12}{[K]} \cdot \underset{12 \times 1}{\{q\}} = \underset{12 \times 1}{\{F\}} \Rightarrow \underset{12 \times 1}{\{q\}} = \underset{12 \times 12}{[K]}^{-1} \cdot \underset{12 \times 1}{\{F\}}$$



$$\underset{18 \times 18}{[K]} \cdot \underset{18 \times 1}{\{q\}} = \underset{18 \times 1}{\{F\}} \Rightarrow \text{REACTIONS}$$

DOF SOLUTION

$$\{q\}_{12 \times 1} = \begin{Bmatrix} u_2 \\ v_2 \\ w_2 \\ \alpha_2 \\ \beta_2 \\ \gamma_2 \\ u_3 \\ v_3 \\ w_3 \\ \alpha_3 \\ \beta_3 \\ \gamma_3 \end{Bmatrix} = \begin{Bmatrix} 0 \text{ mm} \\ -11.94 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0243 \text{ rad} \\ -0.0249 \text{ rad} \\ -0.015 \text{ rad} \\ 29.07 \text{ mm} \\ -31.43 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0269 \text{ rad} \\ -0.0527 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix}$$

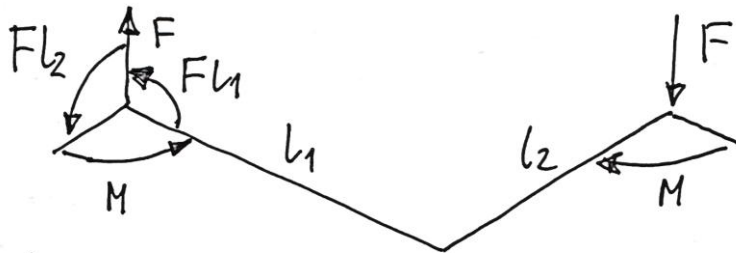


REACTIONS

$$\{F\} = [K] \cdot \{q\}$$

18×1 18×18 18×1

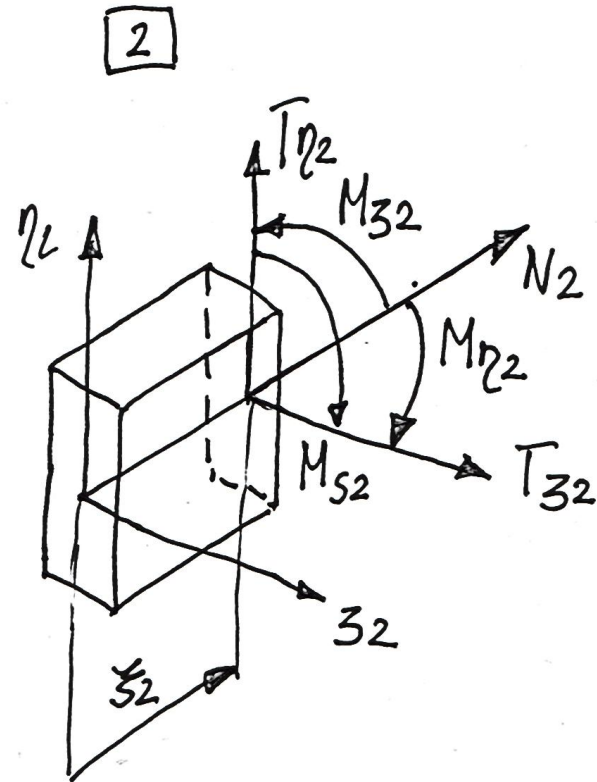
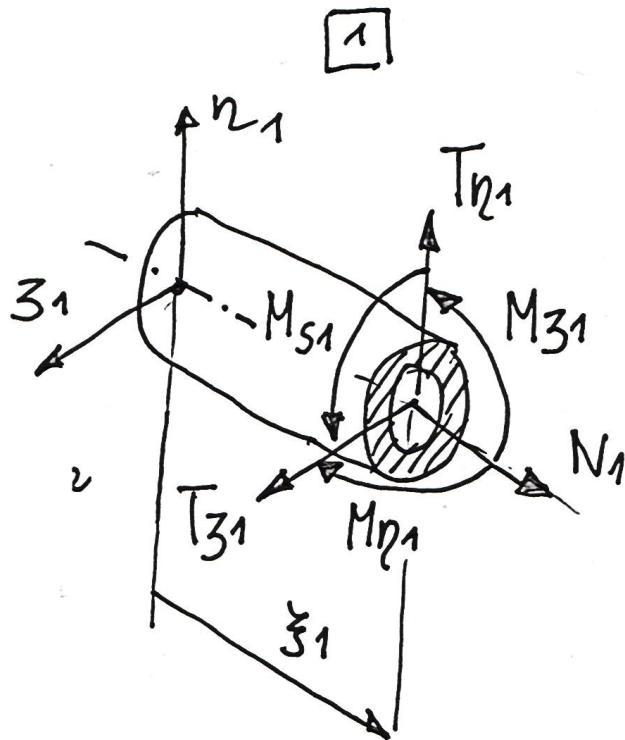
$$= \begin{Bmatrix} R_1 \\ R_2 \\ R_3 \\ M_{ex} \\ M_{ry} \\ M_{rz} \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ -F \\ 0 \\ 0 \\ 0 \\ -M \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 1000 N (+F) \\ 0 \\ 750000 (F \cdot l_2) \\ 1000000 (M) \\ 1200000 (F l_1) \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$



THE STRUCTURE IS IN EQUILIBRIUM

ELEMENT SOLUTION

$$\underbrace{\{q\}_i}_{12 \times 1} = \underbrace{[T_e]_i}_{12 \times 12} \cdot \underbrace{\{q_g\}_i}_{12 \times 1}, \quad i = 1, 2$$



AXIAL BAR : $(q_1, q_7)_i$

$$\varepsilon_{\xi i} = \frac{(q_7 - q_1)_i}{l_i}, \quad \sigma_{\xi i} = E \cdot \varepsilon_{\xi i}, \quad N_i = \sigma_{\xi i} A_i$$

BEAM :

I) BENDING IN $(\xi)_i$ PLANE : $(q_2, q_4, q_8, q_{10})_i$

$$V_i(\xi_i) = \underset{1 \times 4}{[N(\xi_i)]} \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i$$

$$\begin{aligned} M_{\xi i}(\xi_i) &= E \cdot J_{\xi i} \cdot V_i''(\xi_i) = \\ &= E \cdot J_{\xi i} \cdot \underset{1 \times 4}{[N''(\xi_i)]} \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i \end{aligned}$$

$$\sigma_{\xi i}^I = - \frac{M_{3i}(\xi_i) \cdot \eta_i}{J_{3i}}$$

$$\begin{aligned} T_{\eta i}(\xi_i) &= -E J_{3i} \cdot v_i'''(\xi_i) = \\ &= -E J_{3i} \cdot [N_{1 \times 4}'''] \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i = \text{const} \end{aligned}$$

SHEAR STRESS CAUSED BY $T_{\eta i}(\xi_i)$ IS NEGLECTED.

II) BENDING IN (ξ_3) PLANE : $(q_3, q_5, q_9, q_{11})_i$

$$w_i(\xi_i) = \underset{1 \times 4}{[N(\xi_i)]} \cdot \begin{Bmatrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{Bmatrix}_i$$

$$M_{\eta i}(\xi_i) = E \cdot J_{\eta i} w_i''(\xi_i) =$$

$$= E J_{\eta i} \cdot \underset{1 \times 4}{[N''(\xi_i)]} \cdot \begin{Bmatrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{Bmatrix}_i$$

$$\sigma_{\xi i}^{\text{II}} = - \frac{M_{\eta i}(\xi_i) \cdot z_i}{J_{\eta i}}$$

$$\begin{aligned} T_{zi}(\xi_i) &= -E J_{\eta i} \cdot w_i'''(\xi_i) = \\ &= -E J_{\eta i} \left[N_{1 \times 4}''' \right] \cdot \left\{ \begin{matrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{matrix} \right\}_i = \text{const} \end{aligned}$$

SHEAR STRESS CAUSED BY $T_{zi}(\xi_i)$ IS NEGLECTED.

TORSION BAR : $(q_6, q_{12})_i$

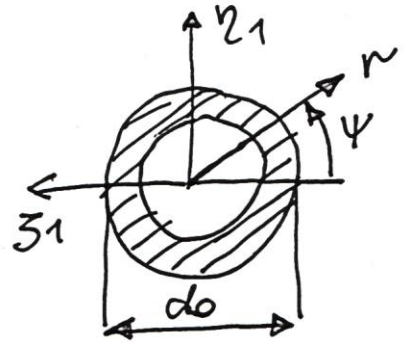
$$\varphi_i(\xi_i) = \underset{1 \times 2}{[N(\xi_i)]} \cdot \left\{ \begin{matrix} q_6 \\ q_{12} \end{matrix} \right\}_i = \left[1 - \frac{\xi_i}{l_i}, \frac{\xi_i}{l_i} \right] \cdot \left\{ \begin{matrix} q_6 \\ q_{12} \end{matrix} \right\}_i =$$

$$= (q_6)_i + \frac{(q_{12} - q_6)_i}{l_i} \cdot \xi_i$$

$$\left\{ \begin{matrix} q_6 \\ q_{12} \end{matrix} \right\}_2 = \left\{ \begin{matrix} 0 \\ 0 \end{matrix} \right\}$$

ELEMENT [1] :

$$\begin{aligned}\tau_1(r) &= G \cdot \gamma_1(r) = \frac{E}{2(1+\nu)} \cdot \frac{d\varphi_1(\xi_1)}{d\xi_1} \cdot r = \\ &= \frac{E}{2(1+\nu)} \cdot \left[-\frac{1}{l_1}, \frac{1}{l_1} \right] \cdot \begin{Bmatrix} q_6 \\ q_{12} \end{Bmatrix}_1 \cdot r = \\ &= \frac{E(q_{12} - q_6)_1}{2(1+\nu)l_1} \cdot r \quad \text{12}, (q_6)_1 = 0\end{aligned}$$



$$\tau_{1\max} = \tau_1\left(\frac{d_0}{2}\right) = \frac{E \cdot (q_{12})_1 \cdot d_0}{4(1+\nu)l_1} = \frac{E d_0 \alpha_2}{4(1+\nu)l_1}$$

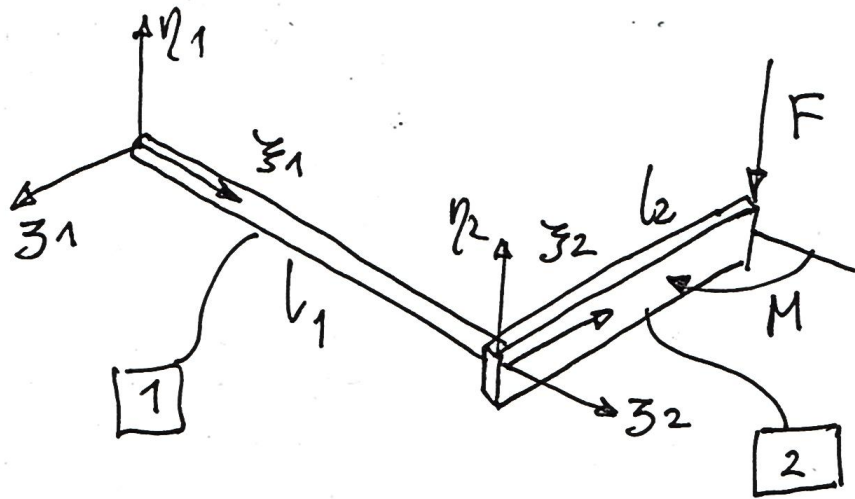
$$M_{s1} = \frac{\tau_1(r) \cdot J_{01}}{r} = \frac{E \cdot (q_{12})_1 \cdot J_{01}}{2(1+\nu)l_1} = \frac{E \alpha_2 J_{01}}{2(1+\nu)l_1} = \text{const}$$

ELEMENT 2 :

$$\begin{Bmatrix} q_{16} \\ q_{12} \end{Bmatrix}_2 = \begin{Bmatrix} \delta_2 \\ \delta_3 \end{Bmatrix} = \begin{Bmatrix} -0.015 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix} \Rightarrow \varphi_2(\xi_2) = q_{16} = \text{const}$$

$$\Rightarrow \frac{d\varphi_2(\xi_2)}{d\xi_2} = 0 \Rightarrow \tilde{\tau}_2 = 0$$
$$M_{s2} = 0$$

ELEMENT RESULTS



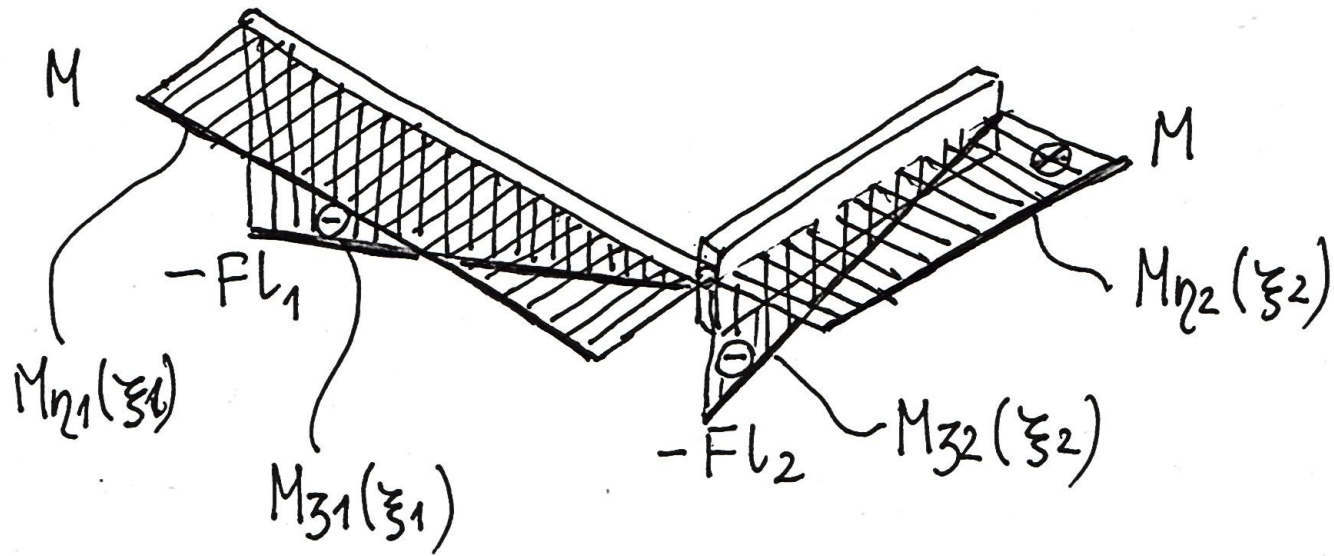
NORMAL FORCES :

$$N_1 = 0, N_2 = 0$$

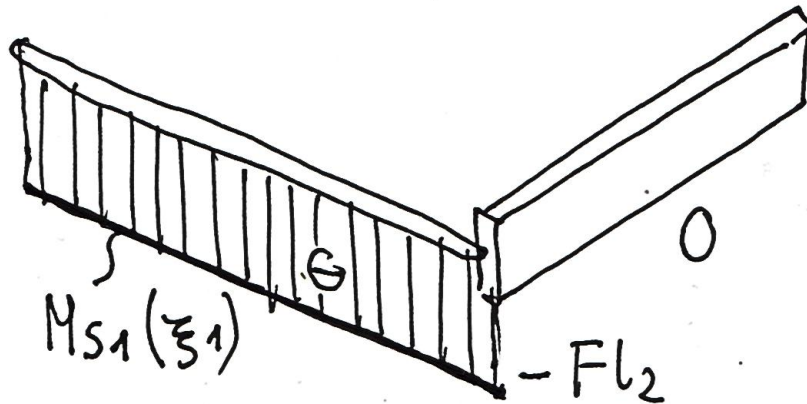
SHEAR FORCES :

$$\begin{aligned} T_{\eta_1} &= -F, & T_{\eta_2} &= -F \\ T_{\zeta_1} &= 0, & T_{\zeta_2} &= 0 \end{aligned}$$

BENDING MOMENTS :



TORQUE. :



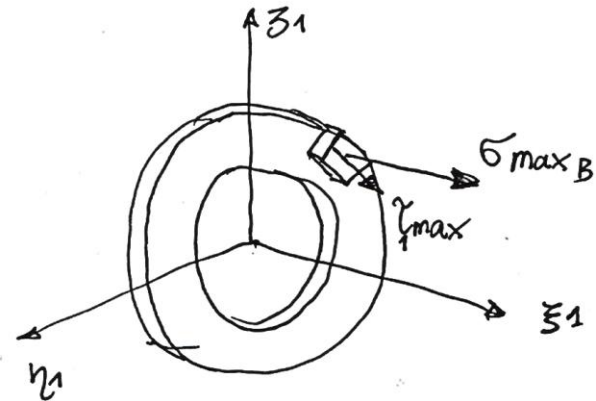
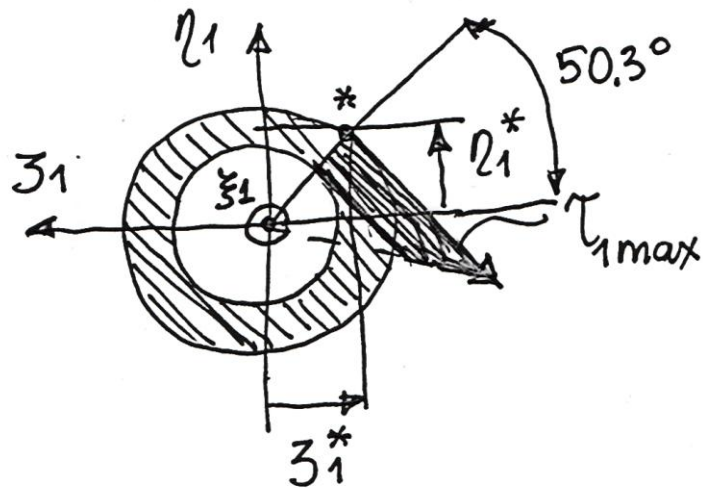
POINT OF THE HIGHEST STRESS :

ELEMENT 1 , $\xi_1 = 0$

NORMAL STRESS DUE TO BENDING :

$$\sigma_{\text{MAX B}} = \sigma_{\xi_1}^{\text{I}} + \sigma_{\xi_1}^{\text{II}} =$$

$$= - \frac{M_{z_1}(0) \cdot \eta_1^*}{J_{z_1}} - \frac{M_{\eta_1}(0) \cdot z_1^*}{J_{\eta_1}} = 161.9 \text{ MPa}$$



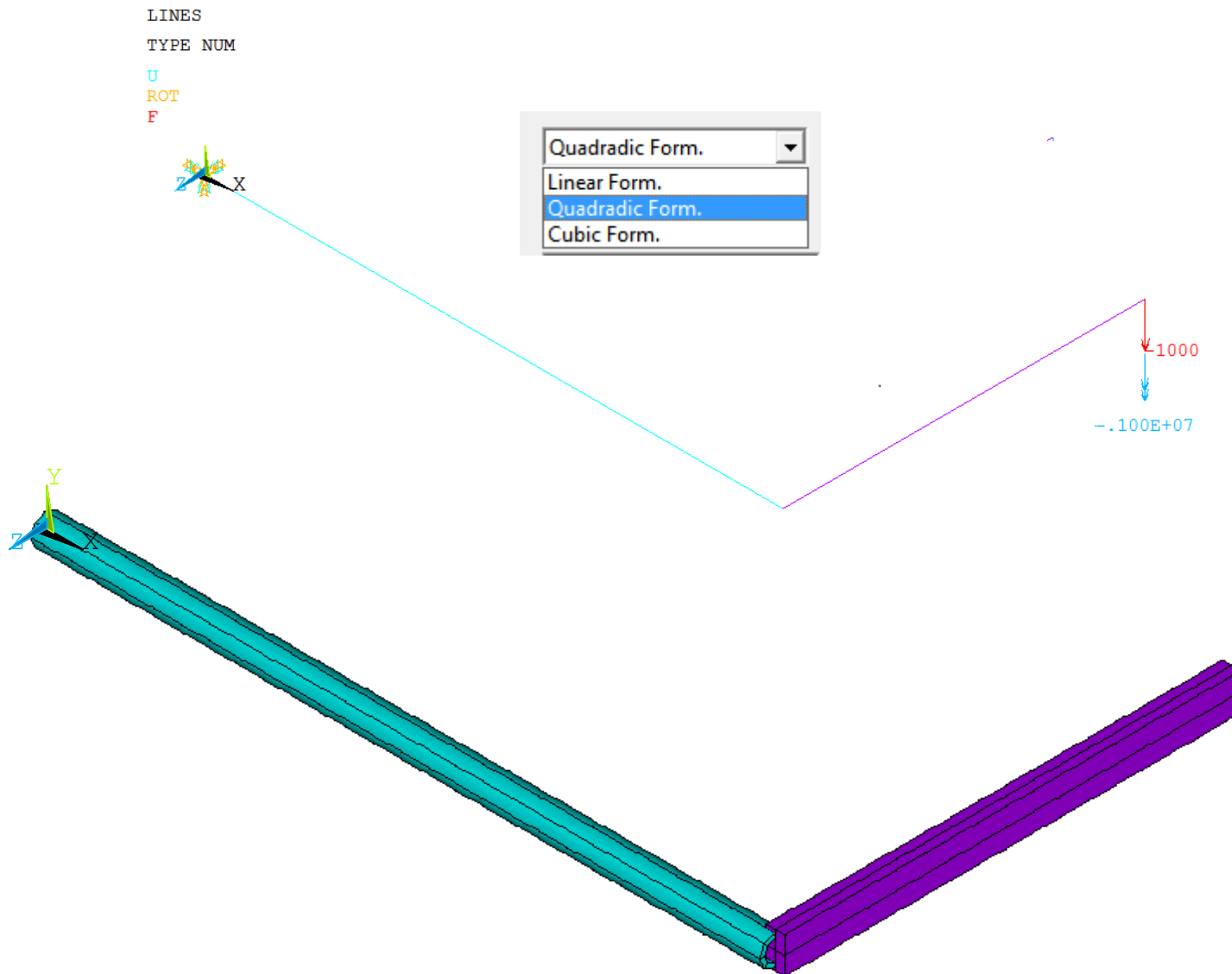
SHEAR STRESS

$$\tau_{1max} = \frac{E d_0 \alpha_2}{4(1+\nu) L_1} = -38.86 \text{ MPa}$$

MAXIMUM EQUIVALENT STRESS :

$$\sigma_{EQU} = \sqrt{\sigma_{MAX B}^2 + 3 \tau_{1max}^2} = 175.34 \text{ MPa}$$

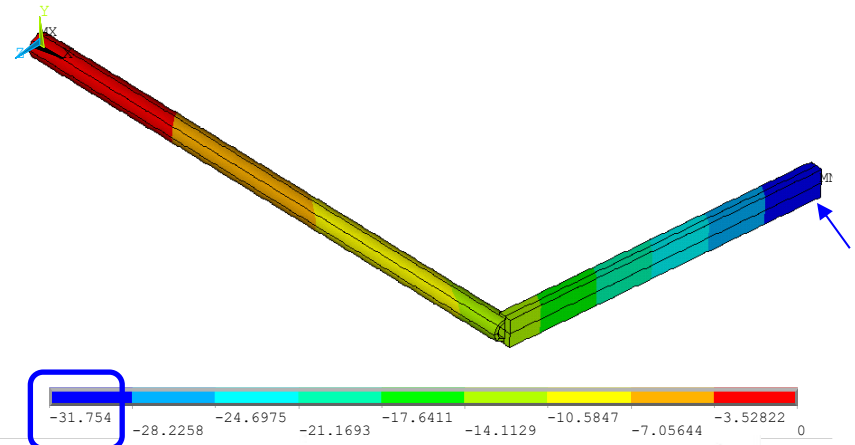
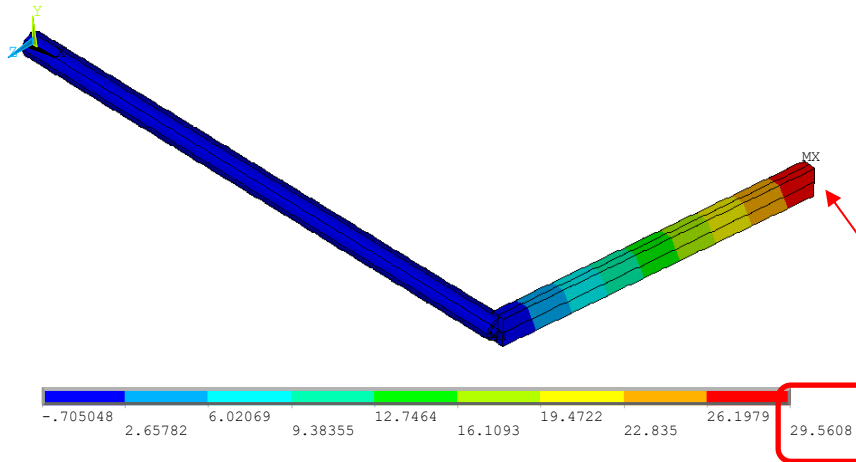
Two-element model (Ansys)



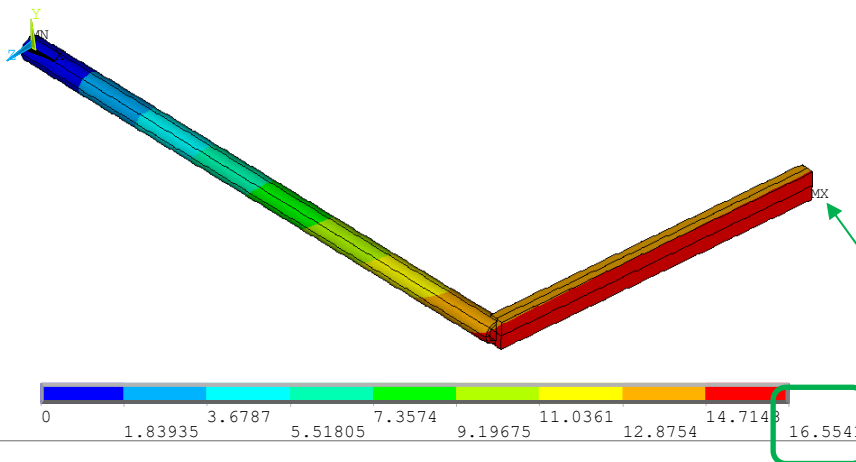
Two-element model (Ansys) – Displacements in X, Y and Z

UX (AVG)
RSYS=0
DMX =45.8841
SMN =-.705048
SMX =29.5608

UY (AVG)
RSYS=0
DMX =45.8841
SMN =-31.754



UZ (AVG)
RSYS=0
DMX =45.8841
SMX =16.5541

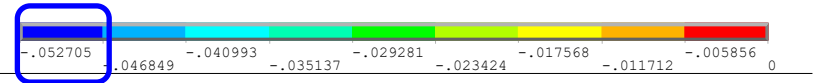
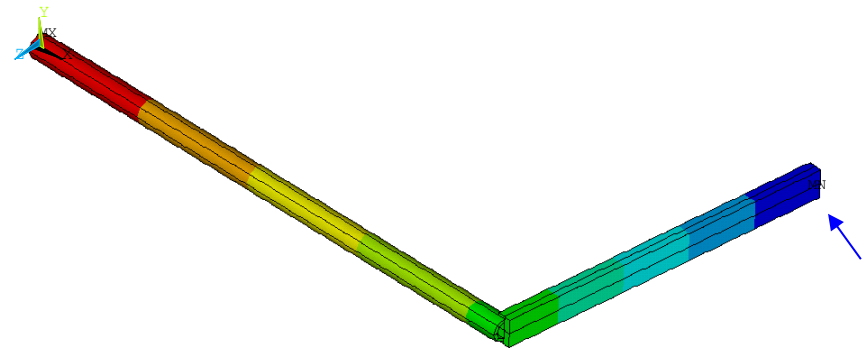
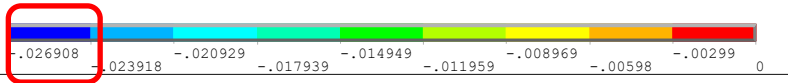
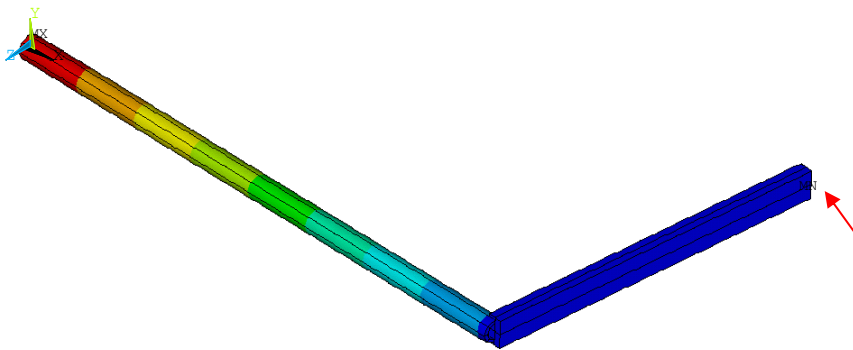


$$\{q\}_{12 \times 1} = \begin{Bmatrix} u_2 \\ v_2 \\ w_2 \\ \alpha_2 \\ \beta_2 \\ \gamma_2 \\ u_3 \\ v_3 \\ w_3 \\ \alpha_3 \\ \beta_3 \\ \gamma_3 \end{Bmatrix} = \begin{Bmatrix} 0 \text{ mm} \\ -11.94 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0243 \text{ rad} \\ -0.0249 \text{ rad} \\ -0.015 \text{ rad} \\ 29.07 \text{ mm} \\ -31.43 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0269 \text{ rad} \\ -0.0527 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix}$$

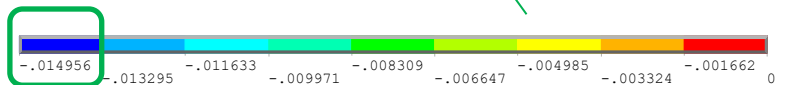
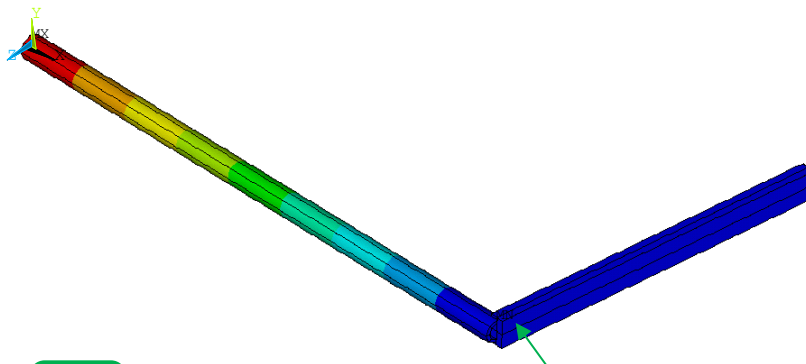
Two-element model (Ansys) - Angular displacements

ROTX (AVG)
RSYS=0
DMX =45.8841
SMN =-.026908

ROTY (AVG)
RSYS=0
DMX =45.8841
SMN =-.052705

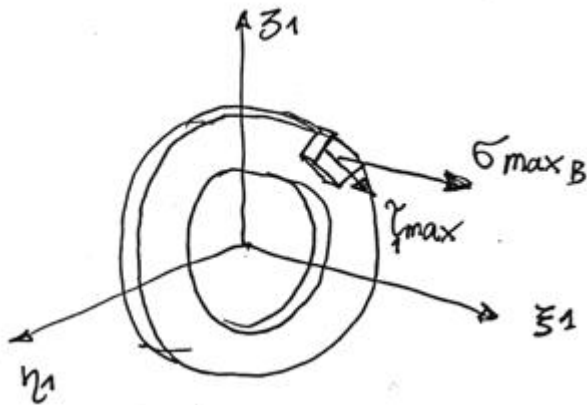
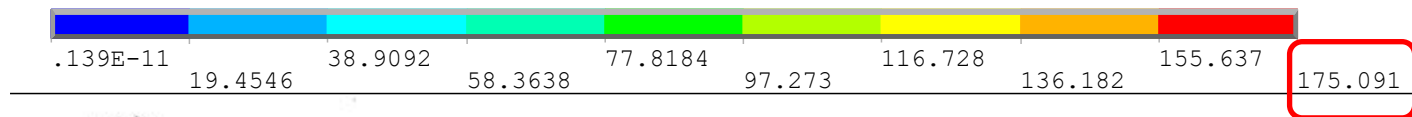
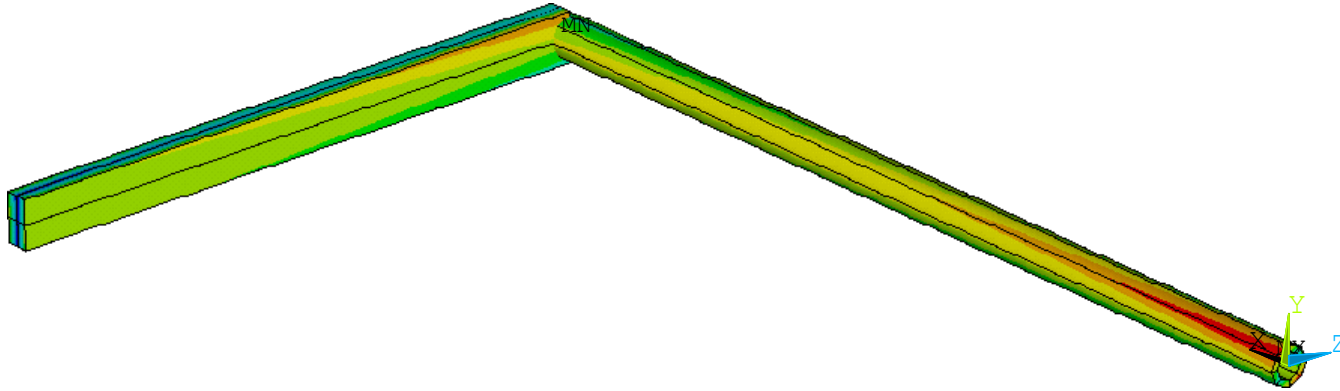


ROTZ (AVG)
RSYS=0
DMX =45.8841
SMN =-.014956



$$\{q\}_{12 \times 1} = \begin{Bmatrix} u_2 \\ v_2 \\ w_2 \\ \alpha_2 \\ \beta_2 \\ \gamma_2 \\ u_3 \\ v_3 \\ w_3 \\ \alpha_3 \\ \beta_3 \\ \gamma_3 \end{Bmatrix} = \begin{Bmatrix} 0 \text{ mm} \\ -11.94 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0243 \text{ rad} \\ -0.0249 \text{ rad} \\ -0.015 \text{ rad} \\ 29.07 \text{ mm} \\ -31.43 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0269 \text{ rad} \\ -0.0527 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix}$$

Two-element model (Ansys) - Von Mises stress (SEQV)



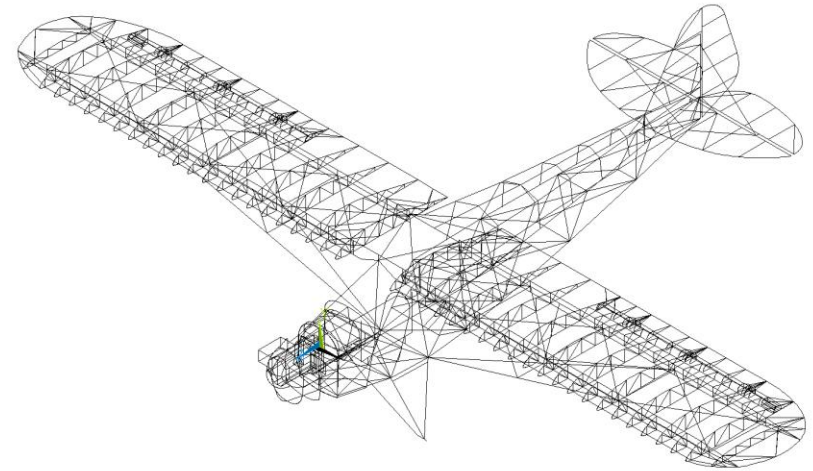
$$\sigma_{EQV} = \sqrt{\sigma_{MAX B}^2 + 3 \tau_{1max}^2} = 175.34 \text{ MPa}$$



LINES
 TYPE NUM

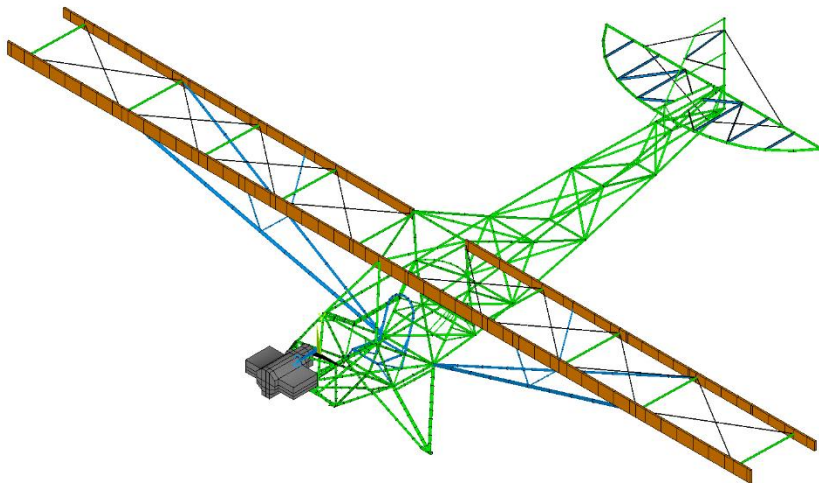
Light aircraft structure (Piper L-4)

ANSYS
 MAY 12 2002
 16:32:00
 PLOT NO. 1

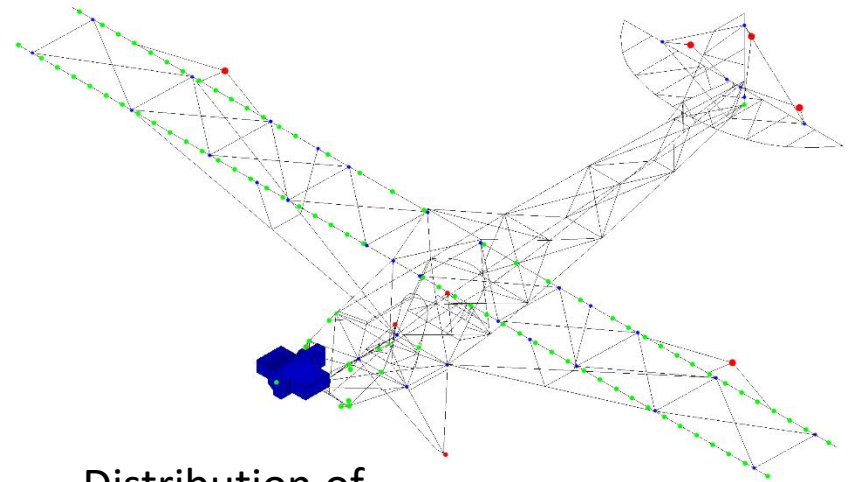


1
 ELEMENTS

ANSYS
 MAY 14 2002
 18:06:41
 PLOT NO. 1



FEM beam model



Distribution of concentrated masses in the substitute model